# An Example of how Capella can be used for CubeSat development

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# **Today's contents**



- Introduction of Speakers
- Why we adapted Capella?
  - What are space systems?
  - What are CubeSats?
  - What we hope to achieve with Capella
- Our Capella model
  - About our CubeSat PRELUDE
  - Software design using Capella
- Conclusion
  - What we have achieved with Capella
  - Future plans

### **Introduction of Speakers**





# Masaki Naito

4<sup>th</sup> year in Nihon University (Undergraduate)

### ■ 6U-sized Satellite PRELUDE Project

- Sub Project Manager
- System Integrator
- On-board software developer
- HEPTA-Sat Training TA

#### **Research Topic**

CubeSat software modeling and optimization by using MBSE including Capella



Satellite Assembly



Laboratory Members

### **Introduction of Speakers**

- Associate Professor in the Department of Aerospace Engineering, Nihon University

- Vice chairman of UNISEC (University Space Engineering Consortium)

#### **Research Topics**

- Large Space Structures (Gossamer Structures)

Masahiko

Yamazaki

UNIVERSITY

NIHON

- Nanosatellite Systems (CubeSats)







Model Order Reduction of Flexible Space Structure Capacity Building Program



Precursory Electric Field Observation CubeSat



### **Research and Project**



#### PRELUDE Satellite Project (will launch FY2025)

- Nanosatellite for observation of earthquake precursor
- Projects that integrate science and engineering through industry-academiagovernment collaboration

### Neptune Satellite Project

- Nanosatellite for early tsunami prediction
- HEPTA-Sat Training
  - Capacity building program through CubeSat hands-on practice



#### Modeling of satellite systems and satellite observation data and solving social problems



Precursory Electric Field Observation





Early Tsunami Prediction Capacity Building for Space Research

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### What are space systems?

A system consists of various elements such as hardware, software, people, data, and services, which work together appropriately to achieve a goal.



- Developing a system means clarifying the boundaries of the system and designing an operational, functional, and physical design that is consistent with the requirements and constraints of the entire life cycle.
- > Evaluated from a variety of perspectives, both numerically and experimentally.



# Payloads do not operate by themselves but rather execute their roles in <u>combination with other on-board hardware, software, and interfaces</u>.

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### What are CubeSats?

**Sizes of Satellites:** The definitions of satellite classes are slightly different from country to country. Generally, large satellites are multi-purpose and use on high accuracy missions. Pico/Nano/Micro satellites may perform similar missions as the large satellites by forming a constellation.



### What are CubeSats?



- The specific standards for CubeSats help reduce costs.
- A 1U CubeSat is a <u>10 cm cube</u> with a mass of approximately <u>1 to 1.33 kg</u>.
- CubeSats come in several sizes, which are based on the <u>standard CubeSat unit</u> (called 1U).



### What are CubeSats?

Nano-satellites that can be developed at low cost and in a short period of time.

- They are being used such as being utilized in education, business, and science.
- Not only replacing conventional technologies, but also complementary to them.



Examples of UNISEC Japan Satellites from 2003 to 2019 (R&D, Commercial)

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# What we hope to achieve with Capella



In the real space project, it is sometimes hard to gain knowledge or experience of the whole development process because the roles are divided among team members.



MBSE provides a visual understanding of other subsystems and the overall satellite system, leading to a design that optimizes the entire satellite.

- Not necessarily created by professionals, so design and verification omissions are likely to occur.
  - Many university students are not familiar with satellite design.
  - Verification omissions can lead to the loss of the satellite.

Drawing and visualizing the model diagram makes it easier to notice omissions.

- In a university institution, knowledge and technology losses occur due to member turnover.
  - High turnover rate (1/3 members graduate every year)
  - By leaving information on the model diagram, it is easier for new members to understand the system.
- It is necessary to develop the system under severe constraints of time, budget, and human resources.
  - We want to optimize satellite system design and development to accomplish our mission with limited resources.

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# **Pre-Earthquake Ionospheric Perturbation**

The small satellite DEMETER was developed by CNES (France) to investigate earthquake precursor in the ionosphere in 2004.



- The mechanism behind these fluctuations remains unknown.
- Multi-aspect statistical evaluations are needed to identify any link with earthquakes.
- There is a shortage of major earthquake cases with high-quality data.

### W6U CubeSat PRELUDE

To increase the number of ionospheric observation data and clarify the mechanism of the phenomena, we are developing the PRELUDE satellite.



# **Three Main Missions of PRELUDE**

1. Detection and statistical evaluation of earthquake precursor



 - 2. Development of 3U size payload for electric field and plasma hybrid observation

- ✓ Only 3U size
- ✓ Simple interface
- ✓ It can also be equipped on other satellites



- 3. Development of CubeSat platform for electromagnetic observation
  - ✓ Reducing electromagnetic noise
  - ✓ Constellation through international collaboration



### **Mission Sequence and Overview**

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### **Arcadia method layers**



#### Analysis is available at different levels of abstraction.



#### Arcadia Method Layers Overview<sup>1)</sup>

1) Hélder Castro: MBSE with Arcadia method step-by-step, <u>https://www.slideshare.net/slideshow/mbse-with-arcadia-methodpdf-256664096/256664096</u>, accessed November 7, 2024



- Main developer of PRELUDE is Nihon University.
  - We are responsible for all system design, development, and operation, from overall satellite design to operation.
- We used Capella as shown in the table below.
- In our case, the satellite system was already designed before we started using Capella.
  - We used the logical layer and physical layer for verification of the satellite system design and optimization of the software.
  - We are still making models in physical layer.

Layer	Objective	
<b>Operational Layer</b>	berational Layer Identifying stakeholders Clarifying the boundaries of the system	
System Layer	Expression of functions required for the entire satellite depending on the mission	
Logical Layer	Expression of functions required for subsystems	
Physical Layer	Expression of functions required for components	

## **Our Capella models: Operational Layer**

- Identifying stakeholders
- Clarifying the boundaries of the system

[OCB] Operational Capabilities



Send Commands from ReartHive data from space

Access Earthquake information



### **Our Capella models: Operational Layer**

- It gives an overall picture of the operation.
- Clarify the "objective" when operating the system.



### **Our Capella models: Operational Layer**

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# **Our Capella models: System Layer**

- Expression of functions required for the entire satellite depending on the mission
- Not only the functions required for the mission, but also the functions to maintain the satellite are necessary.

#### [SAB] System Architecture Blank



**Capella**Days

NI INF 202

- Example of logical layer analysis with breakdown of system layer requirements.
- The analysis stops at the subsystem level elements and does not mention components.



By breaking down the necessary elements for the entire satellite, it is possible to identify the necessary functions of the software without omissions, leading to improved reliability.

You can look at the model of the entire system, or you can focus on individual parts.

### [LAB] Logical Architecture Blank

![](_page_28_Figure_3.jpeg)

#### **Data Handling**

![](_page_28_Picture_6.jpeg)

You can look at the model of the entire system, or you can focus on individual parts.

[LAB] Logical Architecture Blank

**Capella**Days

NI INF 202

![](_page_29_Figure_3.jpeg)

![](_page_30_Picture_1.jpeg)

- We used Capella primarily for software design optimization.
- We consider the following three keywords to be important as indicators of optimization.

### Reliability

- Once the satellite is deployed in orbit, only software can control its behavior.
- Therefore, reliability is needed in satellite software.

### Flexibility

- To deal with unpredictable events in orbit.
- Some things you just don't know until you launch them.

### Reusability

- Different satellites share the same basic functions.
- If common parts can be used, development time can be shortened.

# Software architecture of PRELUDE

- Layered architecture for reuse and maintainability.
- Upper layers are abstracted so that they don't need to change even if hardware changes.
- Application layer is a layer for what the user wants to do.
- Driver layer is a layer for how to do it.

	ayer	Objective	Example
Software (S/W)	Application	Function satisfies requirements for subsystem	Execute command, Monitor satellite health
	Driver	Function to control components	Write data to memory devices, Communication between other OBC
	HAL	Abstract hardware dependencies	SPI transaction, GPIO

# **On-board software development using Capella**

### Logical Analysis (LA)

### Compatible with application layer in software

- How the system behaves to meet demands
- Identify requirements for subsystems

### Physical Analysis (PA)

### Compatible with driver layer in software

- How the system will be built
- Identify requirements for components

![](_page_32_Figure_9.jpeg)

By hierarchizing software, there is no need to change the upper layers even if the lower layers change, thereby improving reusability.

- CapellaDays
- Logical Architecture Blank Diagram focusing only on the acquisition of HK data.
- A program with the same "Function" was created based on this diagram.

![](_page_33_Figure_4.jpeg)

The same functions as in the model are implemented in the software so that we can understand the software visually.

# **On-board software development using Capella**

### 1. Single Command

Command to control satellite movement

### 2. Macro Command

A series of operations can be called up from a single command.

![](_page_34_Figure_5.jpeg)

We find Capella and functional chain useful when designing macro commands, reordering single commands, and considering what is optimal.

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![](_page_35_Picture_1.jpeg)

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### What we have achieved with Capella

- By looking at the model and discussing it, we were able to clarify the requirements for the system.
  - We could logically derive criteria for success/failure of the system.
- By deriving the software module partitioning from Capella, the logic of the software design became clear.
  - The team was able to share the concept of software design.
  - Reliability and reusability of the software were improved.
  - In the future, we would like to apply the Functional Chain concept to software so that we can increase flexibility.

![](_page_36_Picture_7.jpeg)

### **Future plans**

- Now, it is mainly used for software design and success criteria, but in the future, I would like to use it for system design (e.g., for the next satellite).
- I would like to actively utilize Team for Capella, where everyone does the modeling instead of just one person.
  - We would like to make it a common language within the laboratory so that all members can use it.

![](_page_37_Picture_4.jpeg)

https://www.obeosoft.com/en/team-for-capella